## Questions 1-19 refer to either the C-172N/P POH with 180HP/2550LB STC, unless indicated.

- 1. Total usable fuel capacity for the C-172N/P with long range tanks is:
  - a. 54 gallons
  - b. 50 gallons
  - c. 62 gallons
  - d. 40 gallons
- 2. Total fuel capacity for the C-172N/P with long range tanks is:
  - a. 54 gallons
  - b. 43 gallons
  - c. 21.5 gallons
  - d. 50 gallons

3. The maximum certified takeoff weight for the normal category is \_\_\_\_\_ pounds.

- 4. The maximum combined weight capacity for baggage areas 1 and 2 is:
  - a. 100 pounds
  - b. 120 pounds
  - c. 170 pounds
  - d. None of the above
- 5. Maneuvering speeds (KIAS, sea level) are:
  - 2550\_\_\_\_\_ 2150\_\_\_\_\_ 1750\_\_\_\_\_

6. [C-172P Only] Enter the following speeds (KIAS, sea level):

Vx	Vno
Vy	Vne
Vfe (10° flap)	Vglide(no flap)
Vfe (>10° flap)	

- 7. A gradual loss of RPM and eventual engine roughness may result from:
  - a. Formation of carburetor ice
  - b. Loss of oil pressure
  - c. Low fuel
  - d. Magneto problems
- 8. If total loss of oil pressure is accompanied by a rise in oil temperature, there is a good reason to suspect:
  - a. The oil pressure gauge is inoperative
  - b. The outside air temperature is too high for the power setting
  - c. An engine failure is imminent
  - d. The mixture is too lean
- 9. The avionics power switch must be \_\_\_\_\_ during engine start to:
  - a. ON, ensure proper operation of gauges
  - b. ON, ensure the magnetos are operating
  - c. OFF, prevent electrical fire in the engine compartment
  - d. OFF, prevent possible damage to avionics

10. [C-172N Only] During the run-up magneto check, the RPM drop should not exceed

\_\_\_\_\_RPM on either magneto or greater than \_\_\_\_\_RPM difference between magnetos.

11. [C-172P Only] Using 10 degrees of flap for takeoff reduces the ground roll and total distance over an obstacle by approximately \_\_\_\_\_ percent.

- a. 25
- b. 50
- c. 5
- d. 10
- 12. When landing in a strong crosswind, use the following procedure for selecting the approach flap setting:
  - a. Always use 30 degrees of flaps
  - b. Always use 0 degrees of flaps
  - c. Use the minimum flap setting required for the field length
  - d. Use 10 degrees of flaps
- 13. The maximum demonstrated crosswind velocity is \_\_\_\_\_ knots.
  - a. 25
  - b. 10
  - c. 15
  - d. 12
- 14. During a balked landing (go around), reduce the flap setting to \_\_\_\_\_\_ degrees immediately after full power is applied.
  - a. O
  - b. 10
  - c. 20
  - d. 30

15.Using the wind component chart, calculate the wind components for the following conditions:

Runway 19; reported wind 240 at 13 knots

- a. 13K headwind, 17K crosswind
- b. 8K headwind, 10K crosswind
- c. 8K tailwind, 10K crosswind
- d. 10k headwind, 9k crosswind

16. [C-172P Only] Calculate the following takeoff ground roll:

Pressure altitude: 1000 Feet Temp: 30 Degrees C Flaps: Up Weight: 2450 Pounds Wind: 150 Degrees at 12 knots Runway: 19

- a. 961
- b. 1068
- c. 1175
- d. 855

17. Calculate the following cruise performance:

Weight: 2550 Pounds Pressure altitude: 6000ft Temp: 23 Degrees C BHP: 65%

a. 2500 RPM, 114 KTAS, 7.1 GPH

- b. 2400 RPM, 110 KTAS, 8.5 GPH
- c. 2500 RPM, 112 KTAS, 8.8 GPH
- d. 2550 RPM, 114 KTAS, 8.8 GPH

18. Calculate the following short field landing ground roll and over 50 foot obstacle distance using:

Pressure altitude: Sea Level Temperature: 30 Degrees C Flaps: 30 Degrees Weight: 2550lbs Wind: 010 degrees at 10 knots Runway: 19

- a. 666 and 1518 feet
- b. 908 and 2070 feet
- c. 934 and 2130 feet
- d. 818 and 1943 feet

19. [C-172P Only] Calculate the following weight and balance problem:

	Weight	<u> Moment/1000</u>
Basic Empty Weight	1550	57.26
Fuel (50 Gallons)		
Pilot and Front Passenger	340	
Rear Passenger	150	
Baggage Area 1	30	
Baggage Area 2	0	
Ramp Weight & Moment		
Start/Taxi/Run-Up (2 Gallons)		
Takeoff Weight/Moment		

- a. Center of gravity TOO FAR AFT; weight within limits
- b. Aircraft within weight/CG limits in UTILITY category
- c. Aircraft is OVERWEIGHT/CG is within limits
- d. Weight and CG IN LIMITS/NORMAL category

## Questions 20-25 refer to C-172F POH or T-41A Flight Manual.

20. The Normal Category gross weight is \_\_\_\_\_ pounds.

21. The flap extension speed (Vfe) is \_\_\_\_\_ MPH

22. Total usable fuel is \_\_\_\_\_ gallons (standard tanks):

- a. 40
- b. 36
- c. 39
- d. 53

23. The oil capacity is \_\_\_\_\_ quarts and the engine should not be operated with less than \_\_\_\_\_ quarts.

24. The correct fuel management procedure for a VFR flight with a climb to cruising altitude of 5500 feet is:

- a. Fuel selector on BOTH at all times
- b. Fuel selector on BOTH for takeoff and climb
- c. Fuel selector set LEFT or RIGHT during cruise
- d. Both b and c above are correct

25. Enter the following speeds (MPH-sea level)

Vx [C-172F]	Vno [T-41A]
Vy [C-172F]	Vne [T-41A]
Va [T-41A]	Vglide [T-41A] (no flap)

## Questions 26-45 refer to the C-172R POH with 180HP STC.

26. The engine is a Lycoming IO-360 and rated at what horsepower?

- a. 180HP at 2500 RPM
- b. 160HP at 2700 RPM
- c. 180 HP at 2700 RPM
- d. 160HP at 2500 RPM
- 27. The fuel capacity is \_\_\_\_\_ gallons.
  - a. 68 total and 62 usable
  - b. 56 total and 53 usable
  - c. 54 total and 50 usable
  - d. 50 total and 50 usable

28. The minimum operating oil level is 6 quarts. What is the maximum sump oil level?

- a. 8 quarts
- b. 9 quarts
- c. 7 quarts
- d. 6 quarts

29. Compute the landing distance (ground roll) for the following conditions:

PA 1000', 10 Degrees C, Headwind 5 knots

- a. 553 feet
- b. 585 feet
- c. 527 feet
- d. 575 feet

30. The maximum combined weight for baggage areas 1 and 2 is:

- a. 150 pounds
- b. 120 pounds
- c. 170 pounds
- d. Not Defined

31. The Magneto Check is accomplished at 1800 RPM. What is the allowed maximum drop per magneto and the maximum difference between magnetos?

- a. 150 RPM max drop; +/- 50 RPM difference between magnetos
- b. 125 RPM max drop; +/- 50 RPM difference between magnetos
- c. 150 RPM max drop; +/- 25 RPM difference between magnetos
- d. 125 RPM max drop; +/- 25 RPM difference between magnetos

32. The battery is rated at:

- a. 12 volts
- b. 60 AMPS
- c. 45 AMPS
- d. 24 Volts

33. The glide ratio for a C-172 is 9:1. This means that at best glide speed, for every 1000 feet of altitude lost, the distance traveled over the ground is? (assume proper configuration, no wind)

- a. 1 NM
- b. 1.5 NM
- c. 2 NM
- d. 2.5 NM

## (True-A, False-B)

34. (True/False) - To affect an air start with the propeller stopped, the ignition key must be turned to the start position.

35. (True/False) - Alternator malfunction can only be detected by the low voltage warning light.

36. (True/False) - If a total loss of oil pressure is accompanied by a rise in oil temperature, there is good reason to suspect that engine failure is imminent.

37. (True/False) - Use of alternate air can cause a power loss of up to 10% at full throttle.

38. (True/False) - The maximum certificated weight for takeoff is 2550 LBS.

39. (True/False) - The carburetor heat should be pulled at 1500 RPM.

40. The hydroplane speed (9 times the square root of the tire pressure) for the main wheels is?

- a. 55 KIAS
- b. 58 KIAS
- c. 52 KIAS
- d. 60 KIAS

41. The maximum demonstrated cross wind is:

- a. 20 KIAS
- b. 15 KIAS
- c. 16 KIAS
- d. 12 KIAs

42. To lean to the Recommended Lean:

- a. Lean to 50 degrees rich of Peak EGT.
- b. Lean until engine runs rough and then enrich 2 full turns of the mixture knob
- c. Lean to peak EGT
- d. Do nothing
- 43. The annunciator panel provides caution and warning messages for:
  - a. Fuel quantity and Oil Pressure
  - b. Low Vacuum and Low Voltage
  - c. None of the Above
  - d. Both A and B

For questions 44 & 45, compute the takeoff distance at maximum gross weight with the following conditions (C-172R):

Sea Level0 Degrees C2550 lbs6 KT TailwindGrass Surface

44. What is the Ground Roll?

- a. 1118 feet
- b. 1247 feet
- c. 860 feet
- d. 736 feet

45. What is the amount of runway needed to clear a 50 foot obstacle?

- a. 2034 feet
- b. 1594 feet
- c. 1465 feet
- d. 1905 feet